

PROCESS SYSTEMS INTERNATIONAL, INC.					ENGINEERING	NO: V049-1-161
WESTBOROUGH, MA					CALCULATIONS	PAGE 1 OF 13
REV	DEO#	DATE	BY:	CHECK	TITLE:	
0	558	10/17/97	R. Than	D.M.W	TRANSIENT ANALYSIS GATE VALVE WARMUP	
					BY: R. THAN	DEPT: 744
PROJECT: LIGO					PROJECT NO: V59049	
<u>PURPOSE:</u> To determine ramprate to limit the temperature lag between large gate valve gatemass and rest of system to less than twice the steady state value.						
<u>METHOD:</u> Radiation exchange model between adjoining vessels section and valve's gate surface. Transient model						
<u>ASSUMPTIONS:</u> An emmissivity for the gate surface of 0.1 was used For the low emissivity liner an $\epsilon$ of 0.06 was used. The chamber vessel surface: $\epsilon$ of 0.7 The open bore to beam tube/manifold: $\epsilon$ of 0.9 The system surfaces heated directly by the blankets are ramped up at a linear rate until 150°C and held at 150°C.						
<u>INPUTS:</u> GNB DGW: 103227, 103879						
<u>REFERENCE:</u> TIM66, V49-LP-140					<b>OCT 22 1997</b>	
<u>CALCULATIONS:</u> see Attachments						
<u>CONCLUSIONS:</u> Recommend that the ramprate be limited to 1 °C/hr such that the thermal expansion during the transient case is less than twice the steady state case, and ensure that an additional 3 to 4 ft length of insulation, equivalent to 1" fiberglass, is added beyond the heated 3 ft section on the "other side" of the gate. This will improve temperature uniformity along the spool's wall in the angular direction and milder gradient along the axial direction from the heated section to the unheated section.						
<u>NOTES:</u> The emissivity used for the gate surface is conservative, but has less of an effect on the lag for $\alpha=\epsilon$ . The critical parameter is how much length of the beamtube/manifold is heated. This dictates the viewfactor from the gatesurface to the open bore, ambient temperature sink.						

Revision History

Rev.	Comments	Date
0	Release for review	10/10/97

## Simulation Model

For the 44 inch gatevalve, one side of the gate sees a low e liner (cryopump) and the other side sees a short (0.6 m) heated section of the beamtube.

For the 48 inch gatevalve, one side of the gate see a completely heated section and the other side sees a short (0.6m) heated section of the spoolpiece/beammanifold.

An emmissivity for the gate surface of 0.1 was used.

For the low emissivity liner an  $\epsilon$  of 0.06 was used.

For the chamber vessel surface an  $\epsilon$  of 0.7 was used.

For the open bore to beam tube/manifold an  $\epsilon$  of 0.9 was used.

The system surfaces heated directly by the blankets are ramped up at a linear rate until 150°C and held at 150°C.

The time interval used for the each step is 10 seconds.

The net energy to each floating surface was computed using the temperatures of specified ramping surfaces and the current temperatures of the floating surfaces.

The new temperature for each floating surface was computed using the net energy to these surfaces , the heat capacity, and time interval.

The maximum lag occur at the end of the ramp, top of the heating cycle just as the the ramping surfaces reaches the maximum soak temperature.

The lag between the gate surface and rest of the ramping surfaces are:

For 44" gate with the maximum lag is 47°C at a 2°C/hr ramprate.

For 44" gate with the maximum lag is 31°C at a 1°C/hr ramprate

For 48" gate with the maximum lag is 51°C at a 2°C/hr ramprate

For 48" gate with the maximum lag is 34°C at a 1°C/hr ramprate

We should be cautious and proceed with a ramprate that is conservative.

The is some additional mass and surface that needs to be heated, the carriage frame which was not modeled. Therefore, even with a 1°C/hr ramprate, the lag temperature may exceed 34°C.

The following is recommended: ramprate of 1°C and ensure that some additional insulation is added beyond the heated 3 ft section on the "other side" of the gate.

### Thermal expansion

With a 50°C lag the body at the gate opening will be expanded by 0.030" more than the gate surface.

The carriage frame may lag the gate also because of a smaller surface to mass ratio. The thermal expansion difference between the gate and carriage frame is taken up by the linkages. The carriage frame is on wheels. There are two sets of guide wheels, one for each direction.(up/down, sideways). The thermal expansion difference between the body and carriage frame is taken up by the up/down wheels sliding sideways. On warmup the body leads the carriage, so the sideways direction wheels will trail not touch the body walls. On cooldown the gate will lead if the ramprate is small enough otherwise for high enough cooldown rate, at some point the gate surface will lag and the sideways wheels will start to undergo loading from the body.

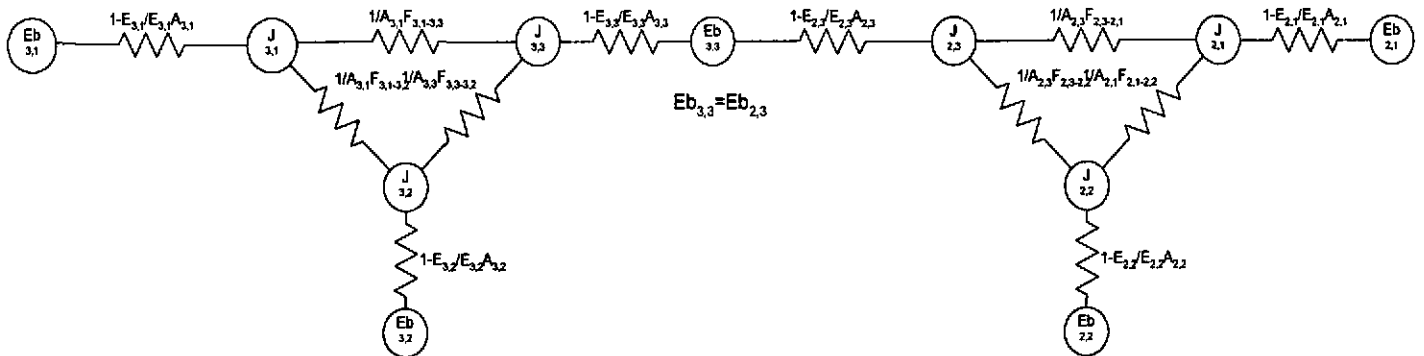
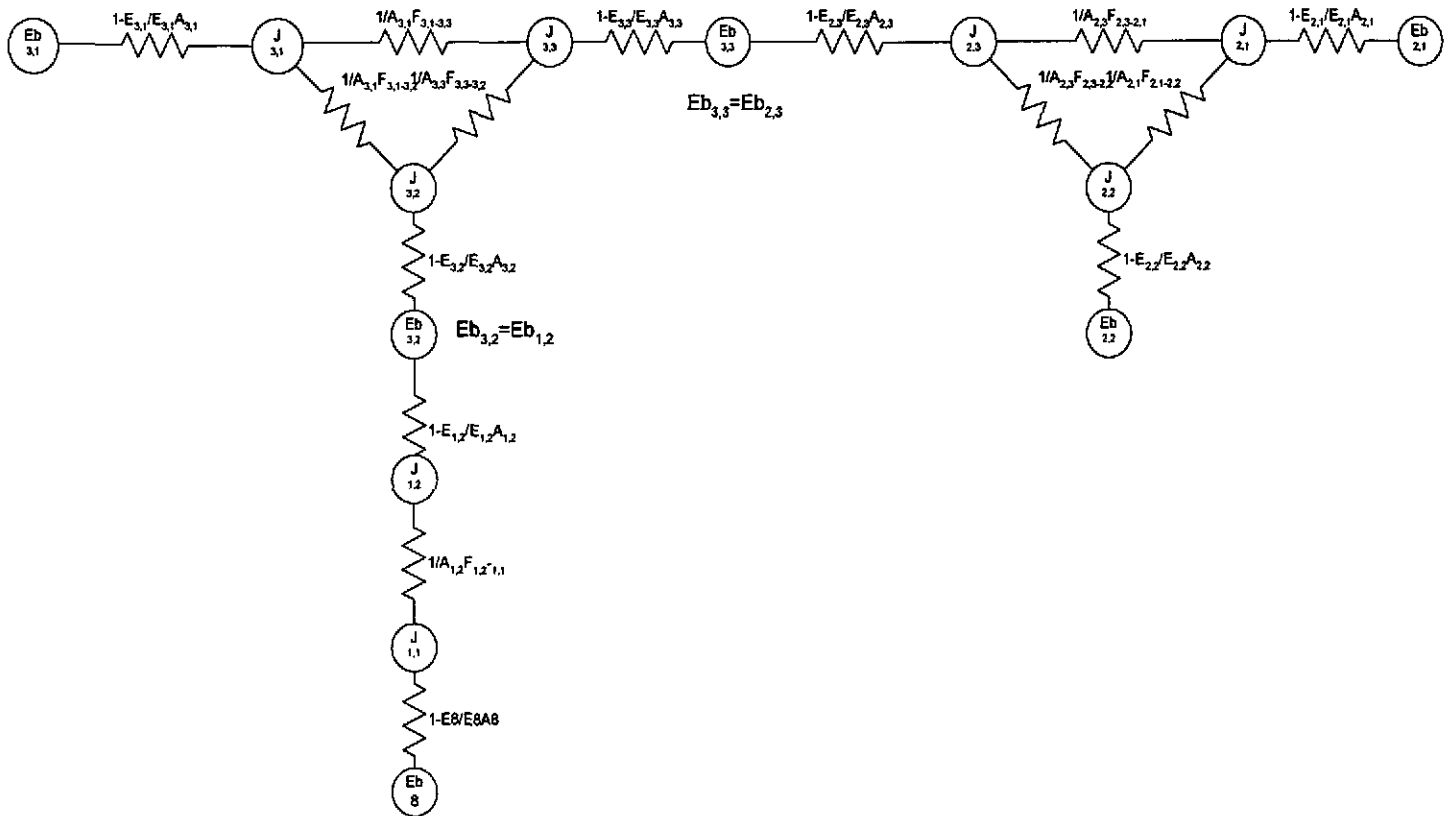
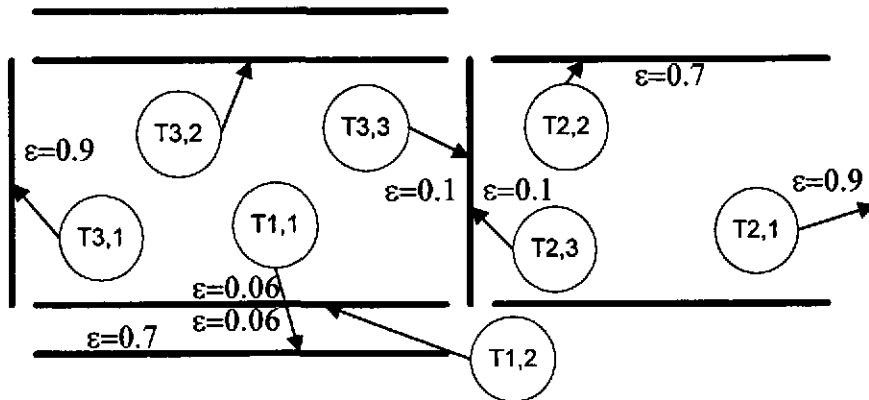


Figure.1



44" Gate Valve with low e liner on one side, Ramprate: 2°C/hr

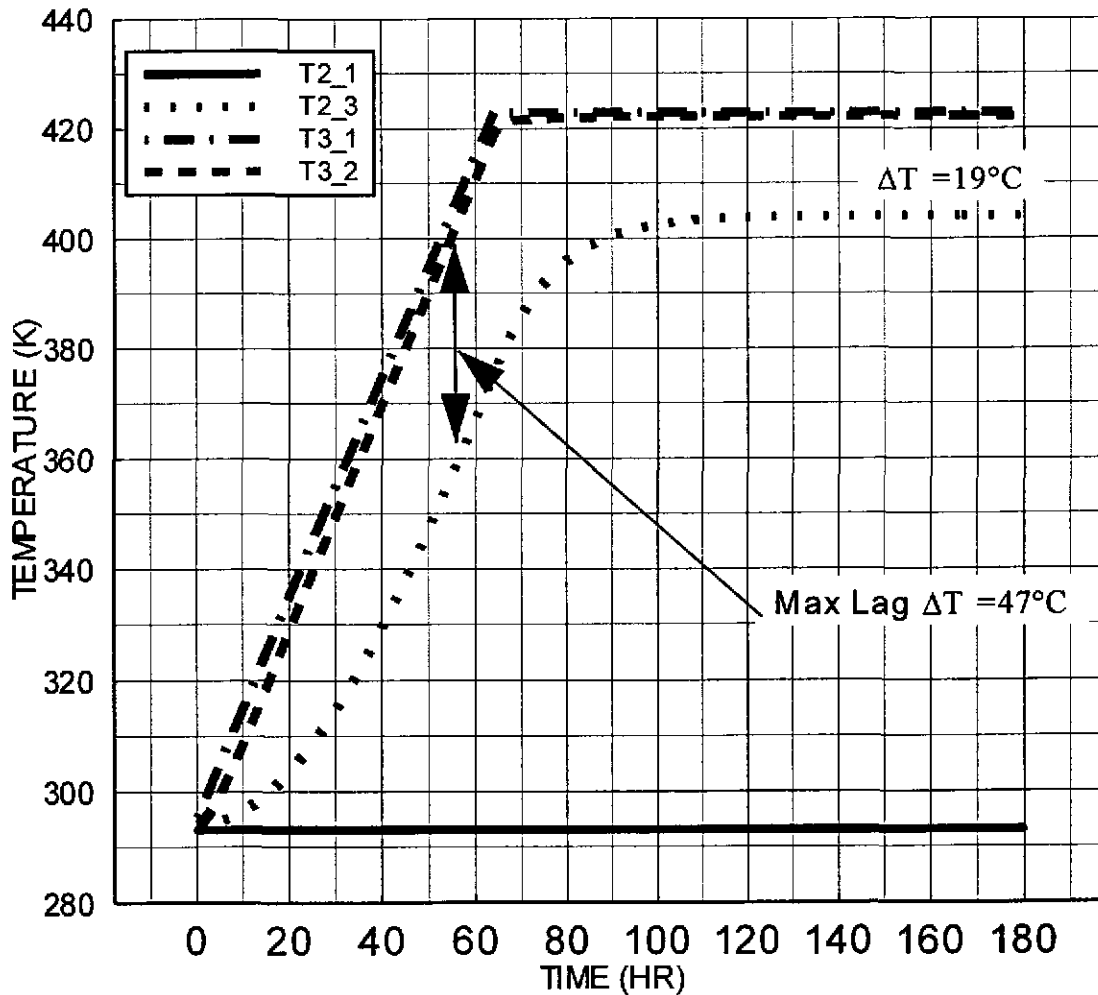


Figure.2

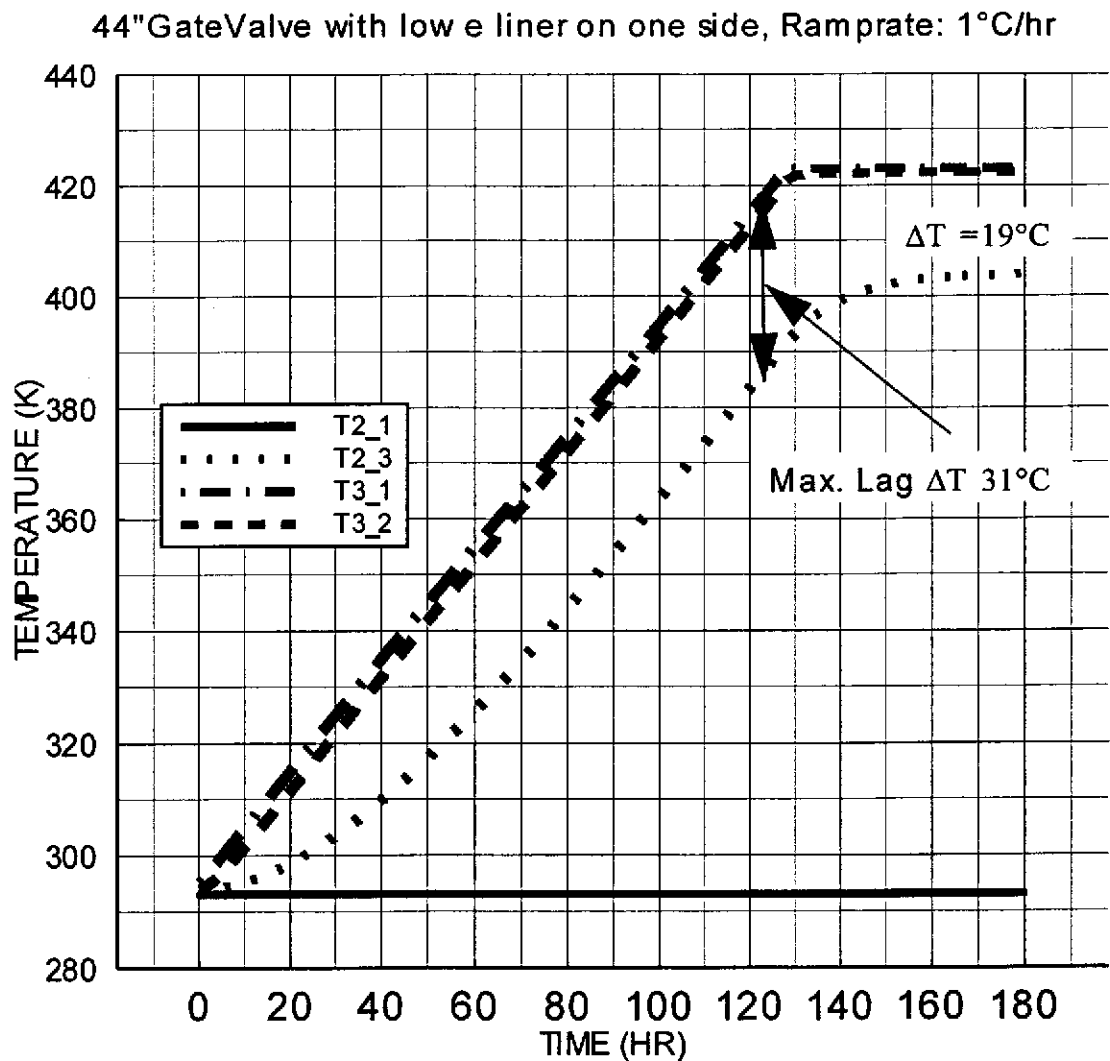
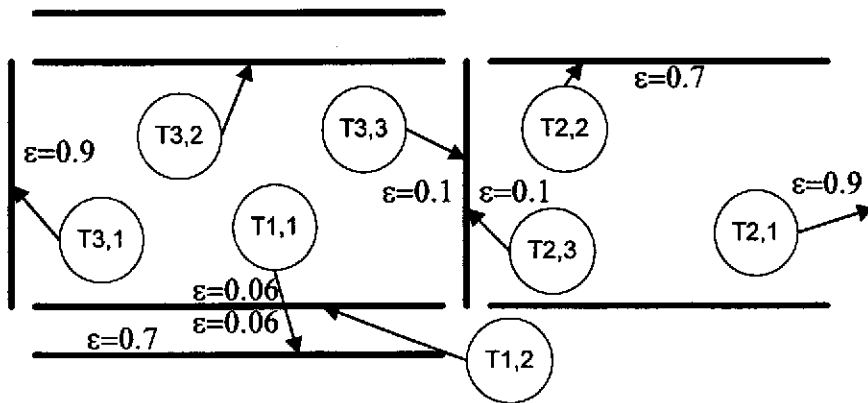


Figure.3

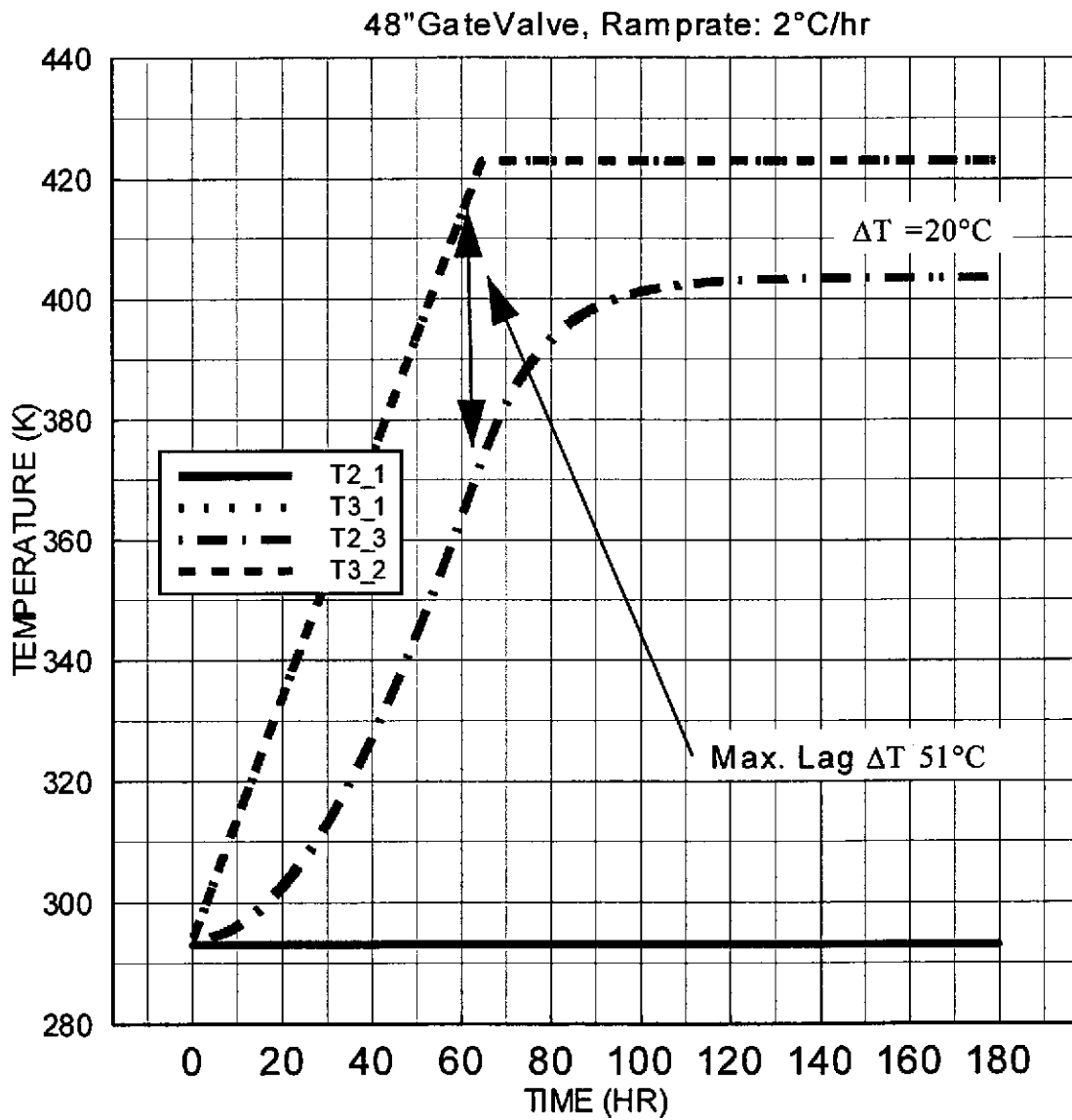
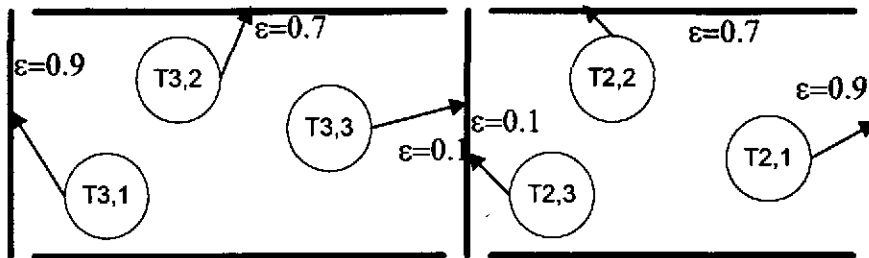
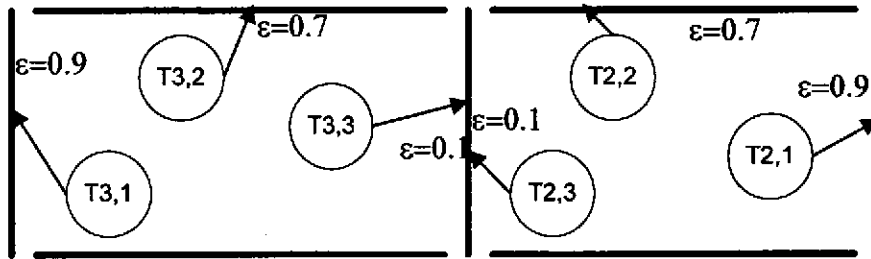
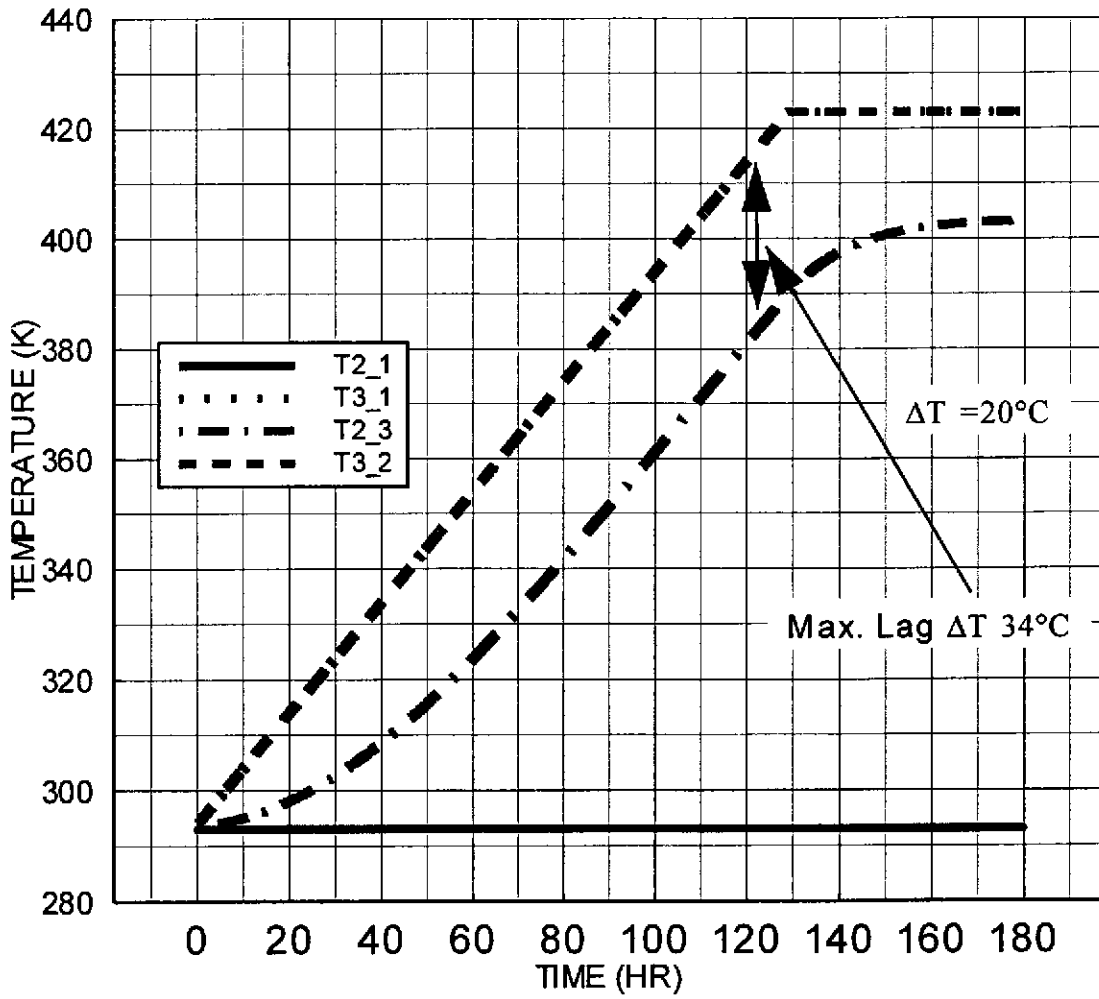


Figure.4



48" Gate Valve, Ramp rate: 1°C/hr





## APPENDIX A

### SIMULATION COMPUTER CODE 44 GATE VALVE

#### C TRANSIENT RADIATION MODEL FOR GATEVALVE WARMUP

```
IMPLICIT DOUBLE PRECISION (A-H,O-Z)
DOUBLE PRECISION A1(100),EM1(100),EB1(100),FV1(100,100),
+RJ1(100),B1(100),AIJ1(100,100),T1(100),Q1(100),QA1(100)
DOUBLE PRECISION A2(100),EM2(100),EB2(100),FV2(100,100),
+RJ2(100),B2(100),AIJ2(100,100),T2(100),Q2(100),QA2(100)
DOUBLE PRECISION A3(100),EM3(100),EB3(100),FV3(100,100),
+RJ3(100),B3(100),AIJ3(100,100),T3(100),Q3(100),QA3(100)
```

C I INDEX NODE/SURFACE , J INDEX NODE/SURFACE IN OTHER DIRECTION SECOND  
SUBSCRIPT

```
C # SYSTEM NUMBER
C A#(I) SURFACE AREA
C EM#(I) EMISSIVITY
C EB#(I) BLACK BODY EMISSIVE POWER
C FV#(I,J) VIEW FACTOR
C RJ#(I) RADIOSITY
C BI#(I) CONSTANT IN LINEAR SET OF EQUATIONS
C AIJ#(I,J) COEFFICIENTS FOR LINEAR SET OF EQUATIONS
```

```
BOLTZ=56.7D-9
PI=3.141592654D0
OPEN(UNIT=8,FILE="G44W2KPH.PRN",STATUS='OLD',ACCESS='SEQUENTIAL',
+ FORM='FORMATTED')
```

C INITIAL CONDITIONS

```
C T1(1) CYLINDER VESSEL WALL BEHIND LOW E SHIELD
C T1(2) LOW E SHIELD SURFACE OUTSIDE DIA
C T2(1) OPEN BORE
C T2(2) CYLINDER VESSEL WALL
C T2(3) GATESURFACE
C T3(1) OPEN BORE SURFACE
C T3(2) LOW E SHIELD SURFACE INSIDE DIA
C T3(3) GATE SURFACE
```

```
T1(1)=295.0
T1(2)=294.0
```

```
T2(1)=293.0
T2(2)=295.0
T2(3)=294.0
```

```
T3(1)=295.0
T3(2)=T1(2)
T3(3)=T2(3)
TIME=0.0D0
DTIME=10.0
```

C SYSTEM 1 PARALLEL CYLINDERS VESSEL WALL / LOW SHIELD PARAMETERS

```
RAD1S1=0.566
RAD1S2=0.566
H1S1=1.0D0
H1S2=1.0D0
```

```
EM1(1)=0.7D0
EM1(2)=0.06
```

FV1 (1, 1)=0.0D0  
FV1 (1, 2)=1.0  
FV1 (2, 1)=1.0  
FV1 (2, 2)=0.0D0

A1 (1)=PI\*2.0\*RAD1S1\*H1S1  
A1 (2)=PI\*2.0\*RAD1S2\*H1S2  
L1=2

C SYSTEM 2 SURFACE 1 CIRCLE, SURFACE 2: CYLINDER SURFACE 3: GATE SURFACE CIRCLE

RAD2S1=0.566  
RAD2S2=0.566  
RAD2S3=0.566  
H2S2=0.60D0

A2 (1)=PI\*RAD2S1\*\*2.0  
A2 (2)=PI\*2.0\*RAD2S2\*H2S2  
A2 (3)=PI\*RAD2S3\*\*2.0

EM2 (1)=0.9D0  
EM2 (2)=0.7D0  
EM2 (3)=0.1D0

C VIEW FACTOR CALCULATIONS

RATIO1=RAD2S1/H2S2

C BOTH CIRCULAR SURFACES ARE SAME SIZE

RATIO2=RATIO1

XX1=1.0D0 + ((1.0 + RATIO1\*\*2.0)/RATIO2\*\*2.0)

FV2 (1, 3)=0.5\*(XX1-SQRT(XX1\*XX1-4.0D0))

FV2 (1, 1)=0.0D0

FV2 (1, 2)=1.0-FV2 (1, 3)

FV2 (2, 1)=A2 (1)/A2 (2)\*FV2 (1, 2)

FV2 (2, 3)=FV2 (2, 1)

FV2 (2, 2)=1.0D0-FV2 (2, 1)-FV2 (2, 3)

FV2 (3, 1)=FV2 (1, 3)

FV2 (3, 2)=FV2 (1, 2)

FV2 (3, 3)=0.0D0

L2=3

C SYSTEM 3 SURFACE 1 CIRCLE, SURFACE 2: LOW SHIELD CYLINDER SURFACE 3:  
C GATE SURFACE CIRCLE

RAD3S1=0.566  
RAD3S2=0.566  
RAD3S3=0.566  
H3S2=1.0D0

A3 (1)=PI\*RAD3S1\*\*2.0  
A3 (2)=PI\*2.0\*RAD3S2\*H3S2  
A3 (3)=PI\*RAD3S3\*\*2.0

C EM3 (1) BORE EMISSIVITY, EM3 (2) SHIELD EM, EM3 (3) GATE EM

EM3 (1)=0.9D0

EM3 (2)=0.06D0

EM3 (3)=0.1D0

C VIEW FACTOR

RATIO3=RAD3S1/H3S2

RATIO4=RATIO3

```

XX3=1.0D0 + ((1.0 + RATIO3**2.0)/RATIO4**2.0)
FV3(1,3)=0.5*(XX3-SQRT(XX3*XX3-4.0D0))
FV3(1,1)=0.0D0
FV3(1,2)=1.0-FV3(1,3)
FV3(2,1)=A3(1)/A3(2)*FV3(1,2)
FV3(2,3)=FV3(2,1)
FV3(2,2)=1.0D0-FV3(2,1)-FV3(2,3)
FV3(3,1)=FV3(1,3)
FV3(3,2)=FV3(1,2)
FV3(3,3)=0.0D0
L3=3

```

C#####

```

C   GATE MASS, HEAT CAPACITY  J/KG-C
C   SHIELD MASS, HEAT CAPACITY J/KG-C
    GATEWT=0.03175
    GATEDIA=1.132
    GATEDENS=7850.0
    SSCP=500.0
    GATEMASS=PI*0.25*GATEDIA**2.*GATEWT*GATEDENS
    GATECP=SSCP*GATEMASS
    SHIELDWT=0.003175
    SHIELDDIA=2.0*RAD3S2
    SHIELDLEN=H3S2
    SHIELDDENS=2700.0
    ALCP=920.0
    SHIELDMASS=PI*SHIELDDIA*SHIELDLEN*SHIELDWT*SHIELDDENS
    SHIELDCP=SHIELDMASS*ALCP

```

C#####

```

C   RAMPRATE
    RAMPCPH=2.0
    RAMPRATE=RAMPCPH/3600.0

```

Cdtddt

C TIME STEP LOOP

```

    WRITE(8,9002)
    DO 700 LTIME=1,64800

```

C=====

C SYSTEM 1 VESSEL CYLINDER WITH LOW E SHIELD CYLINDER

```

C   CALCULATE AIJ1 COEFFICIENTS
    DO 230 I=1,L1
    DO 231 J=1,L1
    IF(I.NE.J) AIJ1(I,J)=-1.0*(1.0-EM1(I))*FV1(I,J)
    IF(I.EQ.J) AIJ1(I,J)=1.0-((1.0-EM1(I))*FV1(I,J))
231 CONTINUE
230 CONTINUE

```

C calculate B1(I) CONSTANTS

```

    DO 202 I=1,L1

    EB1(I)=BOLTZ*T1(I)**4.0
    B1(I)=EM1(I)*EB1(I)

```

202 CONTINUE

```

    CALL gaussedeil(L1,L1,AIJ1,B1,RJ1)
    DO 210 IL=1,L1
    Q1(IL)=A1(IL)*(EM1(IL)/(1.0-EM1(IL)))*(EB1(IL)-RJ1(IL))

```

210 CONTINUE

```

C      WRITE(6,9001)A1(1),A1(2)
C      WRITE(6,9001)EB1(1),EB1(2)
C      WRITE(6,9001)B1(1),B1(2)
C      WRITE(6,9001)RJ1(1),RJ1(2)
C      DO 211 I=1,L1
C      WRITE(6,9001)(AIJ1(I,J),J=1,L1)
211    CONTINUE
C      WRITE(6,9001)Q1(1),Q1(2)
C      WRITE(6,*)" "
C=====
C SYSTEM 2  SURFACE 1 CIRCLE, SURFACE 2: CYLINDER SURFACE 3: GATE SURFACE CIRCLE

C      CALCULATE AIJ2 COEFFICIENTS
        DO 330 I=1,L2
        DO 331 J=1,L2
          IF(I.NE.J) AIJ2(I,J)=-1.0*(1.0-EM2(I))*FV2(I,J)
          IF(I.EQ.J) AIJ2(I,J)=1.0-((1.0-EM2(I))*FV2(I,J))
331    CONTINUE
330    CONTINUE

C      calculate B2(I) CONSTANTS

        DO 302 I=1,L2

          EB2(I)=BOLTZ*T2(I)**4.0
          B2(I)=EM2(I)*EB2(I)

302    CONTINUE

        CALL gaussedeil(L2,L2,AIJ2,B2,RJ2)
        DO 310 IL=1,L2
          Q2(IL)=A2(IL)*(EM2(IL)/(1.0-EM2(IL)))*(EB2(IL)-RJ2(IL))
310    CONTINUE

C      WRITE(6,9001)A2(1),A2(2),A2(3)
C      WRITE(6,9001)EB2(1),EB2(2),EB2(3)
C      WRITE(6,9001)B2(1),B2(2),B2(3)
C      WRITE(6,9001)RJ2(1),RJ2(2),RJ2(3)
C      DO 311 I=1,L2
C      WRITE(6,9001)(AIJ2(I,J),J=1,L2)
C311  CONTINUE
C      WRITE(6,9001)Q2(1),Q2(2),Q2(3)
C      WRITE(6,*)" "

C*****

C=====
C SYSTEM 3  SURFACE 1 CIRCLE, SURFACE 2: LOW SHIELD CYLINDER SURFACE 3: GATE SURFACE
CIRCLE
C      CALCULATE AIJ2 COEFFICIENTS
        DO 430 I=1,L3
        DO 431 J=1,L3
          IF(I.NE.J) AIJ3(I,J)=-1.0*(1.0-EM3(I))*FV3(I,J)
          IF(I.EQ.J) AIJ3(I,J)=1.0-((1.0-EM3(I))*FV3(I,J))
431    CONTINUE
430    CONTINUE

C      calculate B3(I) CONSTANTS

        DO 402 I=1,L3

          EB3(I)=BOLTZ*T3(I)**4.0

```

```
B3(I)=EM3(I)*EB3(I)

402    CONTINUE

      CALL gaussedeil(L3,L3,AIJ3,B3,RJ3)
      DO 410 IL=1,L3
      Q3(IL)=A3(IL)*(EM3(IL)/(1.0-EM3(IL)))*(EB3(IL)-RJ3(IL))
410    CONTINUE

C      WRITE(6,9001)A3(1),A3(2),A3(3)
C      WRITE(6,9001)EB3(1),EB3(2),EB3(3)
C      WRITE(6,9001)B3(1),B3(2),B3(3)
C      WRITE(6,9001)RJ3(1),RJ3(2),RJ3(3)
C      DO 411 I=1,L3
C      WRITE(6,9001)(AIJ3(I,J),J=1,L3)
C411   CONTINUE
      TIME=TIME+DTIME
      TIMEHR=TIME/3600.0
      IF(NPRINT.EQ.60) THEN
      WRITE(8,9001)TIMEHR,T1(1),T1(2),T2(1),T2(2),T2(3),T3(1),T3(2),
+T3(3),Q1(1),Q1(2),Q2(1),Q2(2),Q2(3),Q3(1),Q3(2),Q3(3),DTLAG,QSHLD,
+QGATE
      NPRINT=0
      ENDIF
      NPRINT=NPRINT+1
      QGATE=-Q3(3)-Q2(3)
      QSHLD=-Q3(2)-Q1(2)

      DTRAMP=RAMPRATE*DTIME
      DTGATE=QGATE*DTIME/GATECP
      DTSHLD=QSHLD*DTIME/SHIELDPC
      DTLAG=T2(2)-T2(3)
      T1(1)=T1(1)+DTRAMP
      IF(T1(1).GE.423.0) T1(1)=423.0
      T1(2)=T1(2)+DTSHLD

      T2(1)=T2(1)
      T2(2)=T2(2)+DTRAMP
      IF(T2(2).GE.423.0) T2(2)=423.0

      T2(3)=T2(3)+DTGATE

      T3(1)=T3(1)+DTRAMP
      IF(T3(1).GE.423.0) T3(1)=423.0
      T3(2)=T1(2)
      T3(3)=T2(3)

700   CONTINUE
C*****

9002  FORMAT(8X,"TIMEHR",4X,"T1(1)",5X,"T1(2)",5X,"T2(1)",5X,"T2(2)",
+5X,"T2(3)",5X,"T3(1)",5X,"T3(2)",5X,"T3(3)",5X,"Q1(1)",5X,"Q1(2)",
+5X,"Q2(1)",5X,"Q2(2)",5X,"Q2(3)",5X,"Q3(1)",5X,"Q3(2)",5X,"Q3(3)",
+5X,"DTLAG",5X"QSHLD",5X,"QGATE")
9001  FORMAT(3X,20F10.3)

      END
```