

### **LISA Research and Development**

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### **Contents of This Talk**

- Brief Introduction to LISA
- Interferometry R&D
- Acceleration noise R&D
  ST-7, LTP test flight





### LISA News Flash!

- Beyond Einstein Theme included in Presidents FY 04 Budget
  - LISA ranked first mission. Launch 2011
    - Pending successful TRIP review this spring
    - \$30 M budget for FY04
    - Official Project start this summer
  - Con-X next. Launch 2014

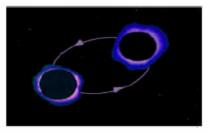


### **The LISA Mission**



#### **LISA Science Goals**

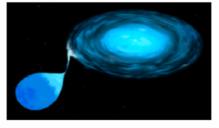
LISA will observe gravitational waves from:



Massive Black Holes; forming from coalescence of seed black holes or from collapse of dense gas clouds, super-massive stars or relativistic star clusters; or coalescing from galaxy mergers



Stellar-mass black holes orbiting massive black holes



Compact binary star systems

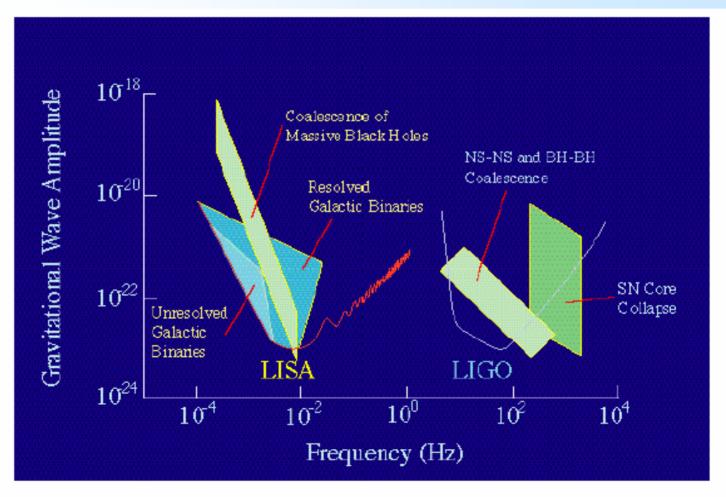
LISA 3



- LISA



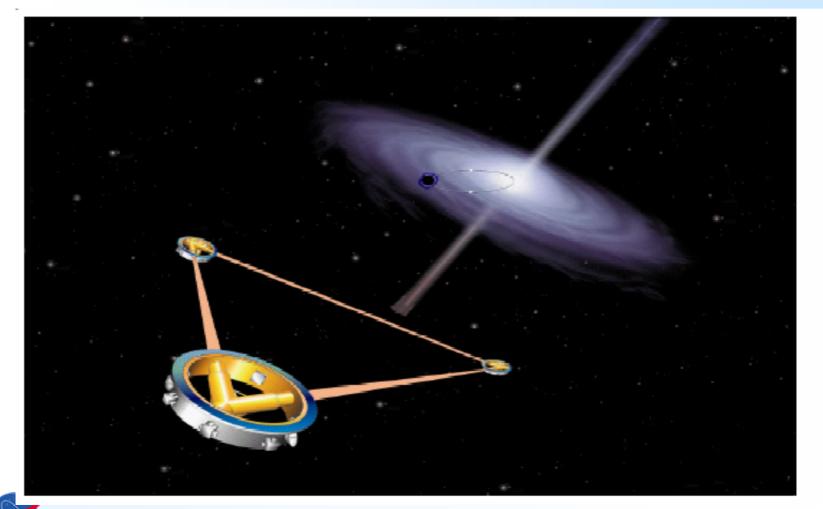
## **Anticipated LISA Signals**





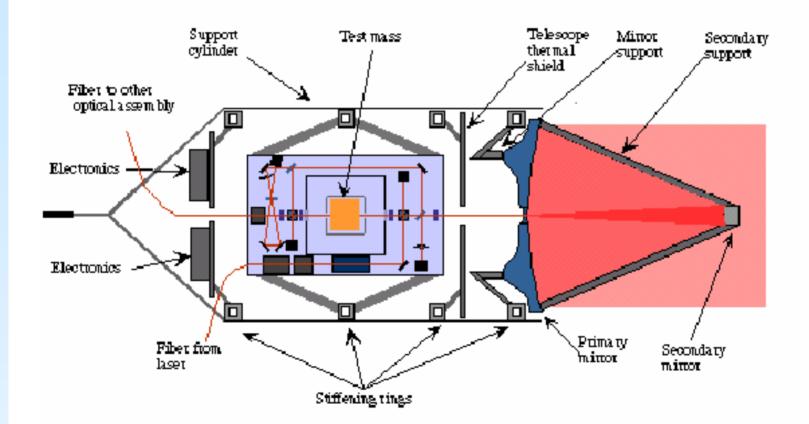


### Mission Concept: Space Michelson Interferometer





### **Optical Layout**





# **LISA Research and Development**

- Interferometry
  - (space) measurement of 10<sup>-12</sup> m over 1000 sec
    - many associated technical issues
- Acceleration noise on test mass
  - 3 x 10<sup>-15</sup> m/sec<sup>2</sup>/sqrt(Hz) over 1000 sec
    - to be validated by ST-7, LTP
- Ground testing of above before launch



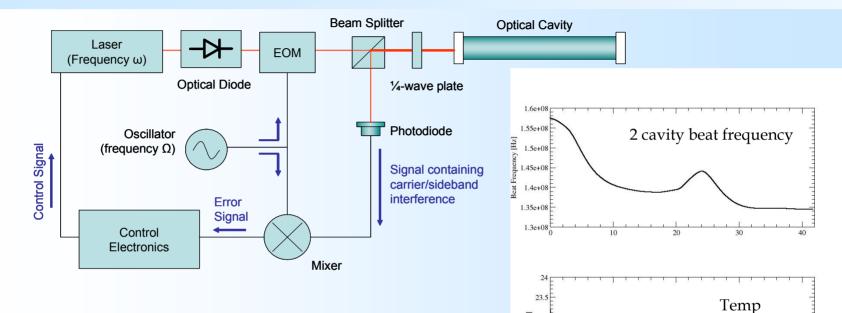
# **Technology of space interferometry**

- Technology of picometer interferometry *in space* 
  - <u>laser stabilization</u>: demonstrate frequency stability of 1 part in 10<sup>14</sup> over 1000 sec timescale
  - <u>materials stability</u>: measure material length stability after launch stress at picometer level
  - <u>signal readout</u>: phase readout of LISA mHz signals in presence of space-induced Doppler shift
  - <u>time delay interferometry</u>: attenuation of frequency noise to 1:10<sup>6</sup> by time-shifting Michelson arm signals





# Laser Frequency Stabilization I Optical Cavity



[emperature [C] 2 22.5

22

21.5 21

- Frequency variation dominated by temperature
  - •1 degree C is ~ 1 nm change for ULE cavity
  - next: temperature stabilization of cavity



20 Time [h]

30

10

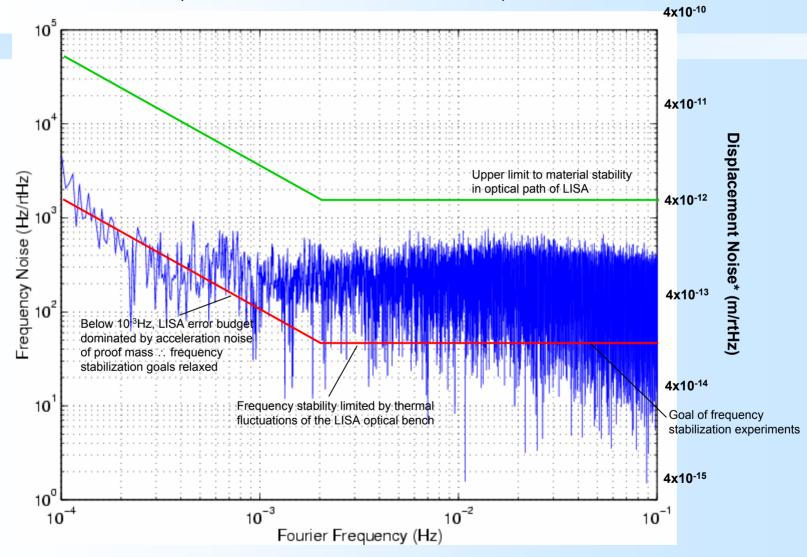


Gold coated cylinders limit radiative transfer to cavity





### Power Spectrum of Beat Note (Muller, McNamara)





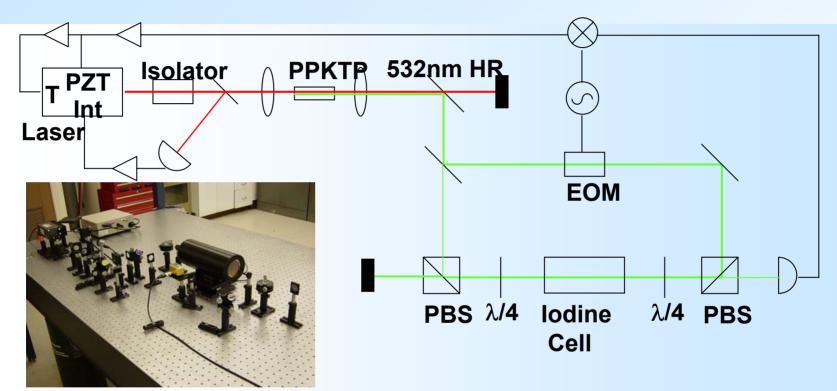


### Laser Frequency Stabilization II: Iodine

- Iodine stabilization of NPRO laser
  - use of iodine atomic transition as frequency reference
    - highly stable (cavity/100 at 10<sup>-4</sup> Hz)
    - fixed frequency, temperature insensitive
    - optical setup more complicated than cavity
- Iodine R&D
  - provides frequency reference for LISA R&D
  - interferometric sensing of LISA and other formation flying missions
    - can we space-qualify iodine setup?



# Iodine Stabilization Optical Layout



- Iodine stabilized lasers are an option for use in LISA
- Iodine stabilized laser will be used to monitor stability of cavity-stabilized lasers
- First results from iodine stabilization expected 1<sup>st</sup> Qtr 2003



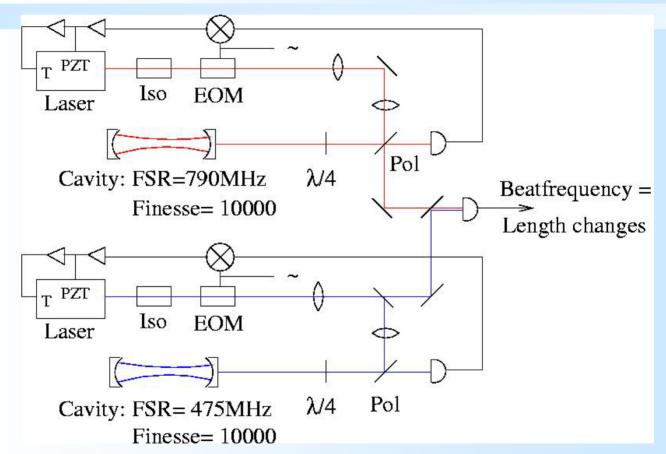


# **Material Stability Studies**

- LISA requires highly stable optical pathlength
  - variety of materials between detectors
    - ULE
    - carbon epoxy
    - optical bonds
  - materials subject to launch stress and vibration
    - stability unknown at picometer level
- research program initiated to study material stability
  - use stabilized laser to track length of stressed material
    - temperature stabilized optical cavity
    - iodine cell



### "Material Investigation" Optical Layout



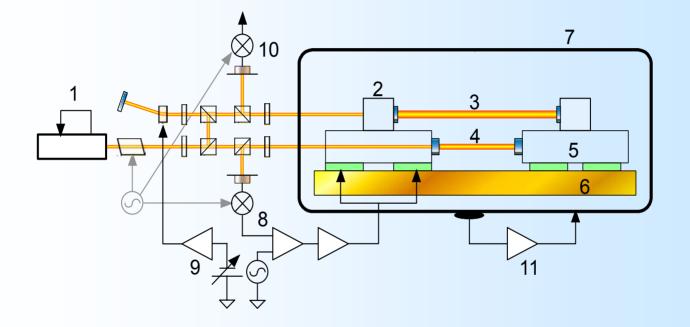
- Each laser locked to own cavity (currently, both cavities made of ULE)
- Light from lasers beat together on photodiode
- Beat note recorded and power spectrum calculated

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GSFC IPI



### **Ground Testing: Control Systems** for Ground Interferometry



•Development of control systems to remove thermal and seismic noise

•auxiliary systems monitor and attenuate noise and provide calibration

• interferometry can then be tested in quiet environment (I&T)





• Acceleration noise < 3 x 10<sup>-15</sup> m/sec<sup>2</sup>/sqrt(Hz) at 1 mHz

### Sources of acceleration noise

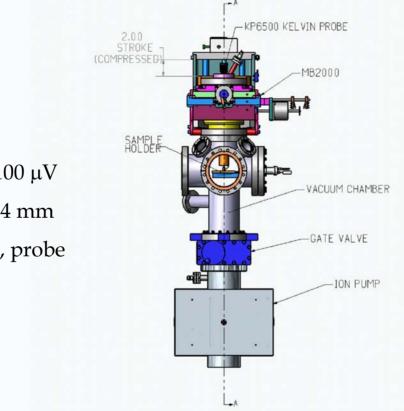
Type of noise	Magnitude	Coupling	Mitigation
Residual gas	0.2	<b>10</b> <sup>-7</sup> <b>torr</b>	Vacuum pump
Radiation pressure	0.3	$\delta T \sim 50 \ \mu K$	Thermal design
Charge fluctuation	0.1	10 e <sup>-</sup> / sec	Discharging
Voltage fluctuation	0.2	$\delta \mathbf{V} \sim 10 \; \mu \mathbf{V}$	Gold coating
Gravity gradient	0.01	$\delta x \sim 10^{-8} m$	Drag-free control
Grav sensor stiffness	0.4	$\delta x \sim 10^{-8} m$	Drag-free control

### • ST-7, LTP will test understanding of these noise terms





### **Mother of all Kelvin Probes: Study of Surface Voltage Noise**

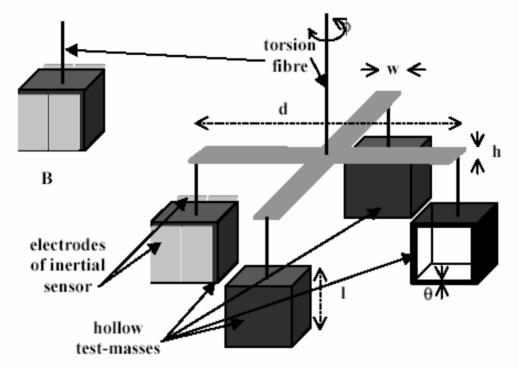


- Voltage sensitivity 100  $\mu$ V
- Spatial sensitivity 0.4 mm
- Gold coated sample, probe

• Delivery of Probe in March

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### Acceleration Noise Measurements: Torsion Pendulum



- Thermal noise limit of 10<sup>-13</sup> nt/sqrt(Hz) at 5 mHz
  - Grav sensor stiffness: S. Vitale (U. Trento)
  - Surface voltage variation: J. Gundlach (U. Washington)
- possible improvement of sensitivity with fused silica fiber



- Technology Development 2003 2006
- SM2 flight in 2006
  - NASA and ESA test packages
  - validation of noise models
  - downselection of project technologies
- Design and Fabrication 2006 2010
- Integration and Test 2010 2011
- Launch 2011

