Torsion Pendulum Ground Testing Results for LISA Gravitational Reference Sensors

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5th Amaldi Conference, Tirrenia July 8th 2003

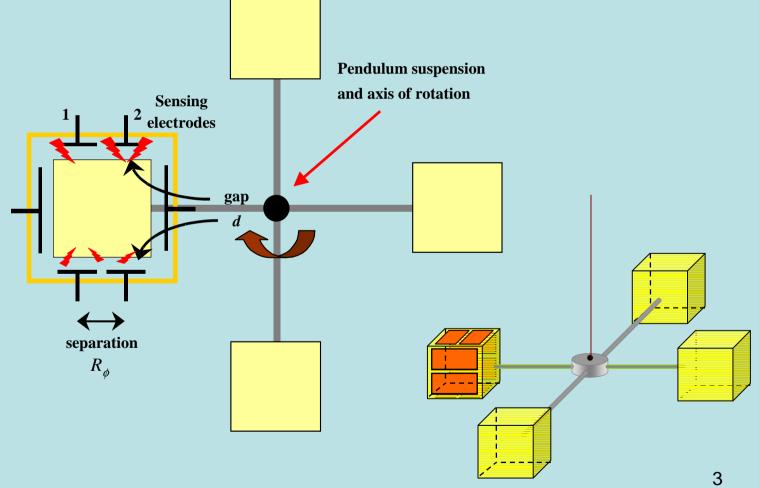
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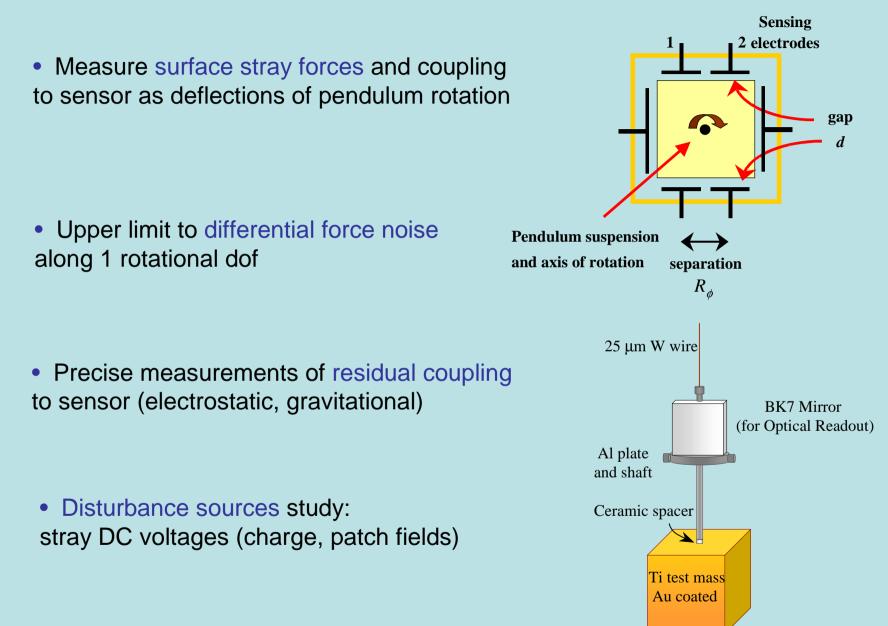
Talk summary

- n Scheme of the experiment
- n Random force noise measurements
- n Sensor disturbances characterization:
 stray DC bias measurements and compensation electric charge measurements electrostatic coupling (stiffness)
- n Status and future of the facility

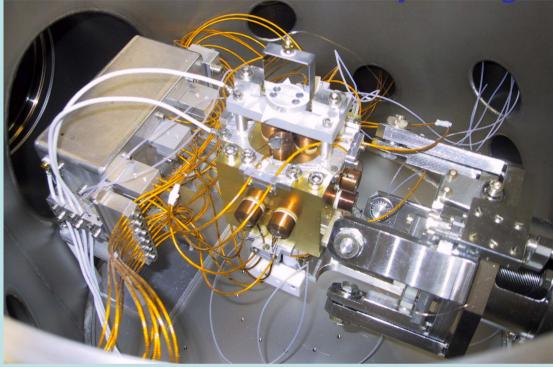
Testing LISA gravitational sensors with torsion pendulum

• Light weight test mass suspended as part of the inertial member of a low frequency torsion pendulum, surrounded by representative LISA sensor housing

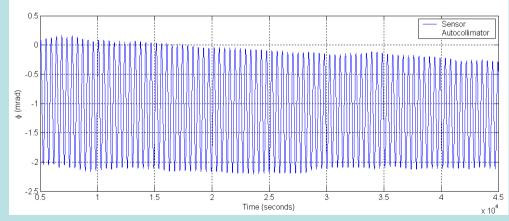




The facility at a glance

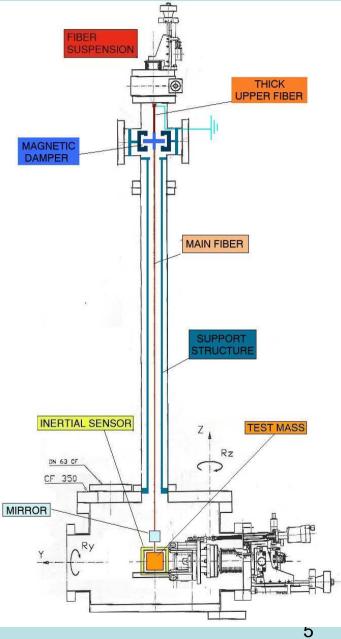


25 μm 1m W fiber



 $T_0 = 513 \text{ s}$

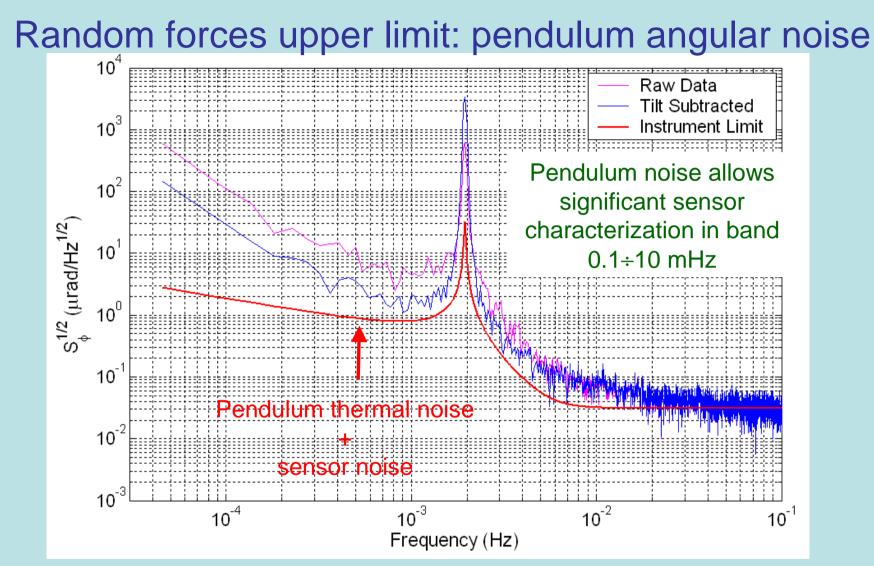
Q = 1750



The facility at a glance



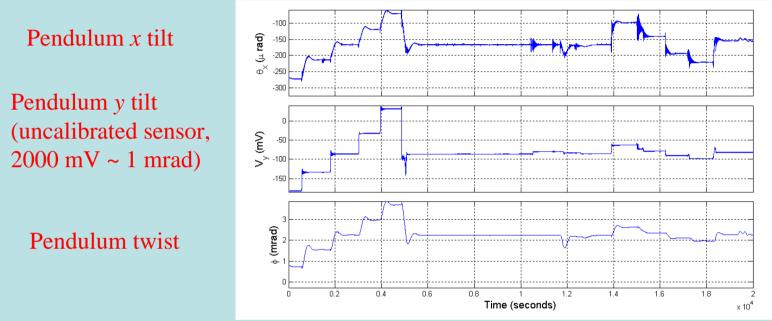
Many more details in poster by L.Carbone *et al*



- Angular noise of roughly 6 times the thermal limit at 1 mHz
- Tilt-subtracted data by measured coupling coefficients: roughly 2-3 times the thermal limit

Pendulum tilt noise

- "Tilt-twist" feedthrough couples tilt into torsional mode
- Measured: ~ 300 nrad/Hz^{1/2} at 1 mHz

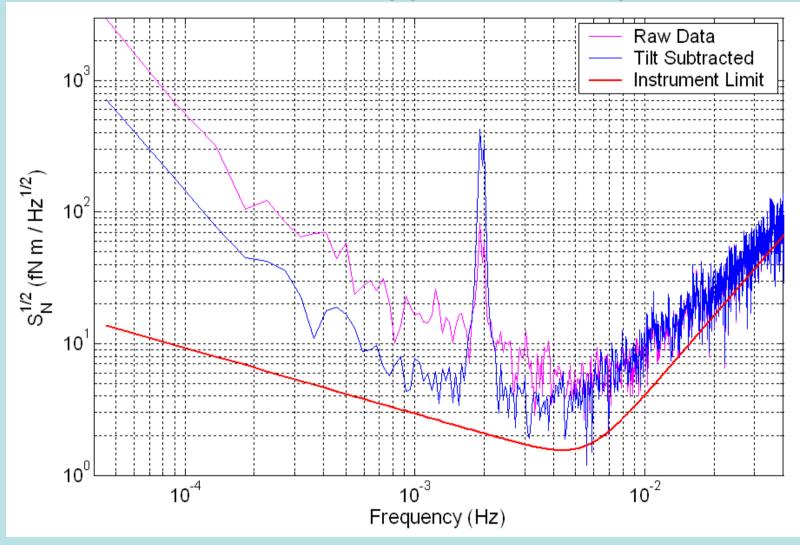


- Tilt pendulum platform, measure tilt and resulting pendulum rotation
- Time series subtraction of effect of floor tilt, with sensor θ_x, θ_y data and measured coefficients

$$\Delta \phi(t) = T_{\theta_x} \Delta \theta_x(t) + T_{\theta_y} \Delta V_Y(t)$$

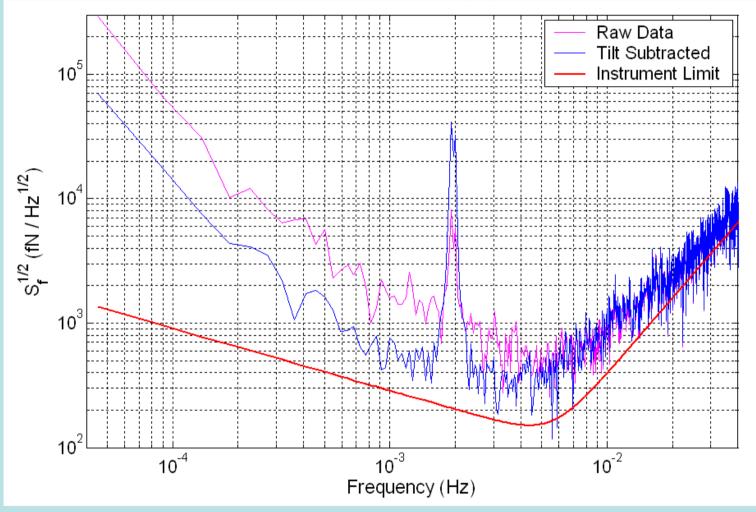
$$\phi(t) = \phi_{raw}(t) - \Delta \phi(t)$$

Random forces upper limit: torque noise



• Torque noise <10 fN m/Hz^{1/2} between 0.6 and 10 mHz

Random forces upper limit

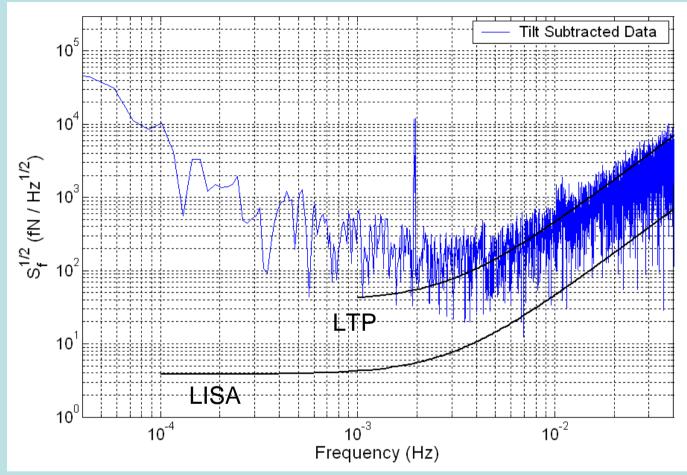


Conversion from torque noise to differential force noise:

• Sensor back-action: arm length L = $R_{\phi} \approx 10 \text{ mm}$

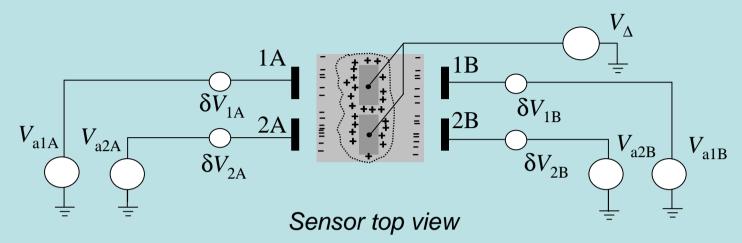
Randomly distributed surface effects: L ≈ 20 mm

Impact for LISA



- Upper limit on random surface forces: molecular collisions, homogeneusly distributed patch charges
- Relatively insensitive to bulk gravitational/magnetic disturbances and to net forces

Electrostatic torque measurements: modulated DC biases

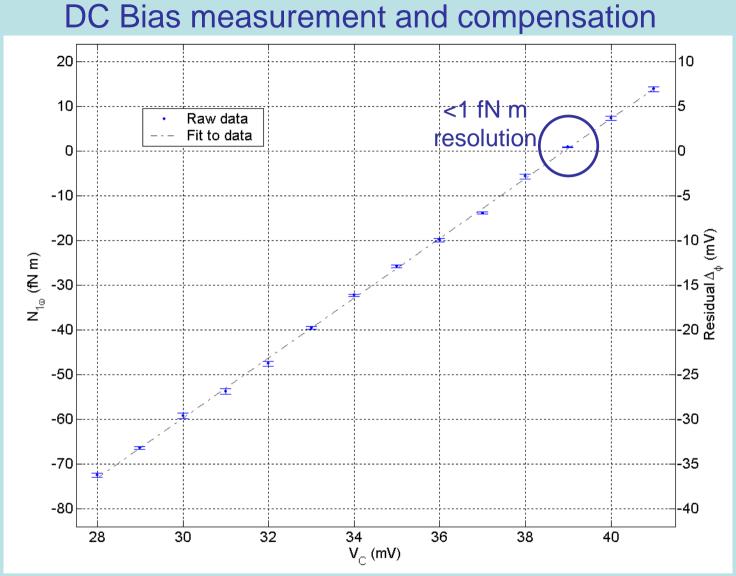


- Modulate test mass voltage with V_{Δ} = 3V amp at f_0 = 5 mHz on injection electrodes
- Charge and DC biases produce torques at voltage driving frequency
- Measured $1\omega_0$ torque sensitive to "diagonal" DC bias imbalance

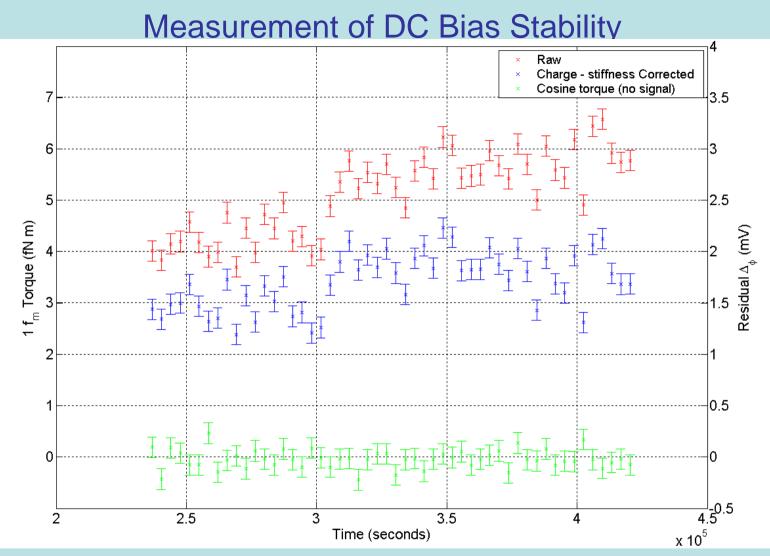
$$\Delta_{\phi} = \delta V_{1A} + \delta V_{2B} - \delta V_{1B} - \delta V_{2A}$$

• Compensate with applied voltages:

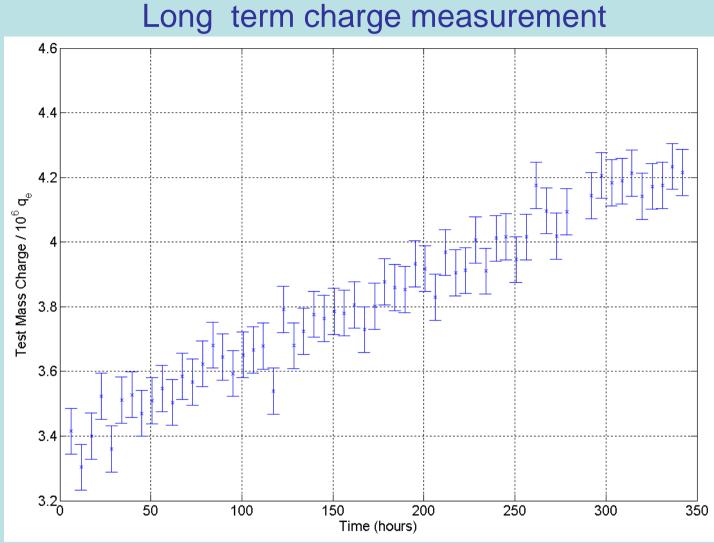
$$V_{a1A} = V_{a2B} = -V_{a1B} = -V_{a2A} \equiv V_{COMP} = -\Delta_{\phi} / 4$$



- DC Bias imbalance measured and compensated within 1 mV
- Problem for LISA: 10 mV level

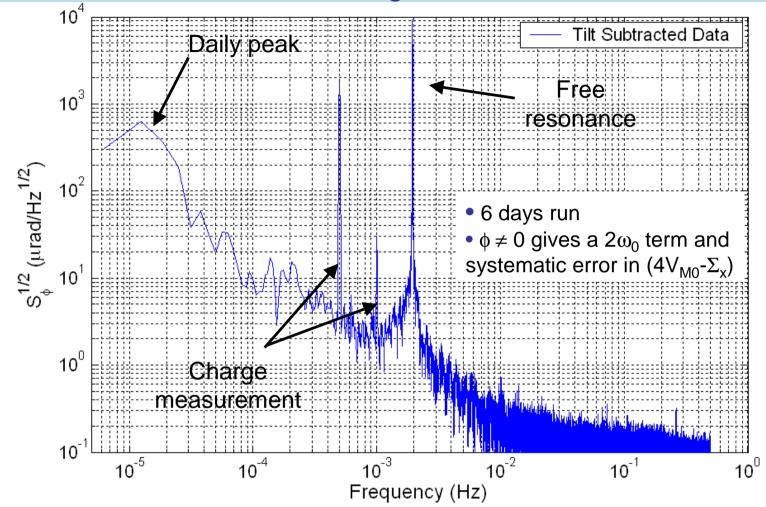


- DC bias imbalance stable to within 1 mV over 50 hour run
- Measurement noise (cosine phase) is below the observed scatter in the electrostatic signal



- Roughly +1 e/s charging
- Measurement error of 5.10⁴ charges in a 6 hour measurement with 50 mV measurement bias

Test mass charge measurement



Ground-based demonstration of the technique to be used in flight for:

- Stray DC fields
- Test mass charge

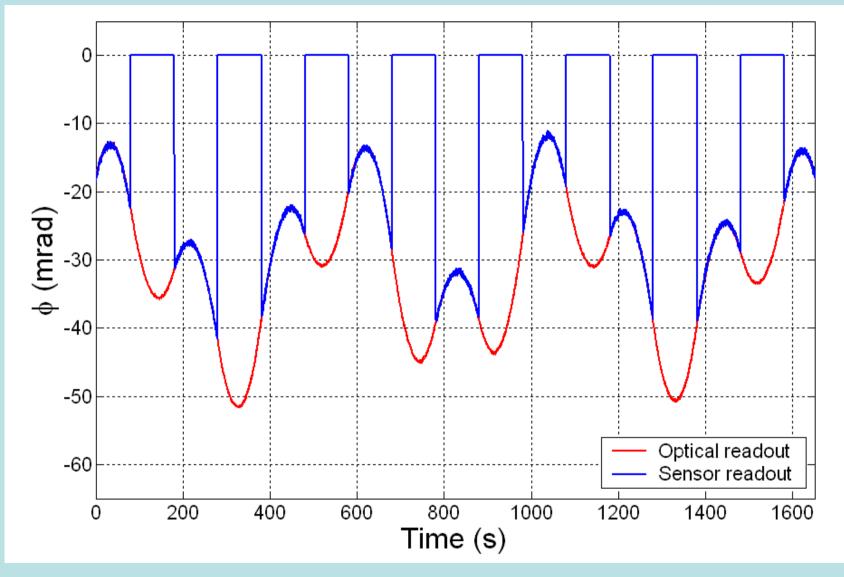
Measurement of electrostatic sensing stiffness

- Sensor bias voltage produces electrostatic rotational stiffness V_{INI} $N_0 = \frac{1}{4} \left(V_{in0} \frac{C_{in}}{C_{TOT}} \right)^2 \sum \frac{\partial^2 C}{\partial \phi^2} (\phi - \phi_0) = -\Gamma_{sens} (\phi - \phi_0)$
- Squarewave modulation of 100 kHz sensor bias amplitude produces squarewave torque N(t)

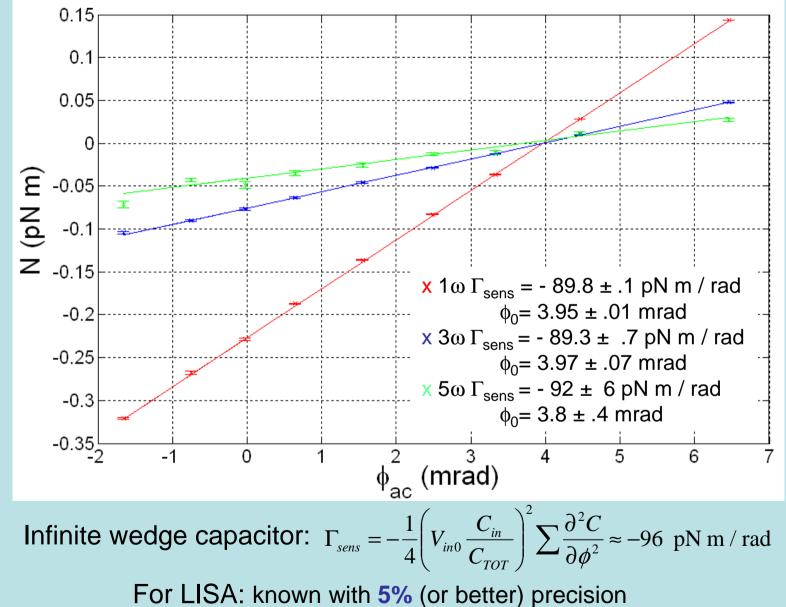
$$N(t) = N_0 \times \left\{ \frac{1}{2} + \frac{2}{\pi} \sin \omega_0 t + \frac{2}{3\pi} \sin 3\omega_0 t + \frac{2}{5\pi} \sin 5\omega_0 t + \dots \right\}$$

- Coherent detection of $1\omega_0$, $3\omega_0$ torques ... each give independent estimate of squarewave torque amplitude
- Measure torque $N(\phi)$ as function of position to get the torque gradient or angular stiffness: $\Gamma_{sens} = -\partial N_0 / \partial \phi$

Electrostatic stiffness measurements: coherent torque scheme



Electrostatic stiffness measurements: position dependence



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Status of the facility:

- "natural end" of the experiment due to fiber detaching from pendulum
- 6 months continuous operation

Next future:

- mirror and inertial member Au coating to reduce chargig effects
- charge management system device for LTP (breadboard)

Future:

- multiple mass configuration to study translational forces
- improved tilt immunity
- ...

Expermental results in submitted paper, on line at:

 $http://xxx.lanl.gov/PS_cache/gr-qc/pdf/0307/0307008.pdf$