# Tip-Tilt Mirror Suspension: Beam Steering for Advanced LIGO Sensing and Control Signals

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We describe the design of a small optic suspension system, referred to as the Tip-Tilt mirror suspension, used to isolate selected small optics for the interferometer sensing and control beams in the Advanced LIGO gravitational wave detectors. The suspended optics are isolated in all 6 degrees of freedom, with eigenmode frequencies between 1.3 Hz and 10 Hz. The suspended optic has voice coil actuators which provide an angular range of  $\pm 4$  mrad in the pitch and yaw degrees of freedom.

Keywords: gravitational wave detectors, interferometer, steering mirrors, suspension, isolation

## I. INTRODUCTION

Second generation interferometric gravitational wave detectors, such as Advanced LIGO<sup>1</sup>, are currently under construction. These detectors use optical metrology to achieve displacement sensitivities of  $\sim 10^{-20}$  m/ $\sqrt{\rm Hz}$  at 100 Hz, with a signal bandwidth from 10 Hz to 10 kHz. The detectors use large 34 cm diameter suspended optics to form a Michelson interferometer with 4 km long Fabry-Perot cavities in each arm. The gravitational wave signal is obtained via a 'DC-readout' scheme<sup>2</sup>. This scheme registers small power fluctuations on DC coupled readout photodiodes at the output of the interferometer . In order to prevent non-TEM<sub>00</sub> mode light and light in the RF control sidebands from reaching the readout photodiodes, a short (~1 m) Fabry-Perot cavity is placed just in front of the readout photodiodes. This 'output mode cleaner' is typically a monolithic cavity, mounted in a suspension for vibration isolation. Mode-matching and beam steering optics are required to direct the optical beam from the output of the large suspended interferometer into the suspended output mode-cleaner. Reflective optics are used to provide actuation for an active alignment control system. In addition to steering the beam into the output-mode-cleaner, suspended optics are required for beam steering of other interferometer sensing ports such as the 'reflected port' and 'pick-off port'. To reduce any coupling of vibrational motion into the gravitational wave readout, the optics must be isolated in all 6 degrees of freedom. In this paper, we discuss the design of such a suspended steering mirror, the Tip-Tilt Mirror Suspension (or Tip-Tilt for short), which will be used in the Advanced LIGO detector.

The mechanical design and the horizontal and vertical isolation of the Tip-Tilt mirror suspension are described first. This is followed by a discussion of local sensing and actuation. We conclude by presenting the overall isolation and steering performance.

#### II. MECHANICAL DESIGN

The Tip-Tilt mirror suspension has been designed to provide isolation of a 2" diameter mirror in all 6 degrees of freedom, and control of 3 of them. All the eigenfrequencies of these 6 degrees of freedom are designed to be below 10 Hz, the lower end of the gravitational wave signal band. The quality factors of the eigenmodes of the Tip-Tilt mirror suspension are damped to values less than 100. This reduces the sensitivity (around the resonances) to external motion of the Tip-Tilt mirror suspension as a whole.

The Tip-Tilt mirror suspension is designed around an aluminium ring, which holds a standard 2"  $\varnothing \times 3/8$ " thick (50.8  $\times$  9.52 mm) optic. Mechanical isolation of the suspended aluminium ring is provided by means of a single stage pendulum and cantilever blade springs. A schematic of the suspension is shown in

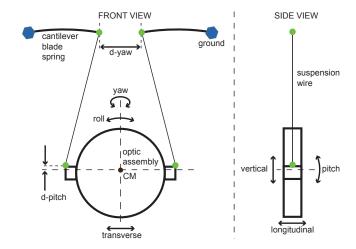


FIG. 1. Schematic overview of the suspended optic with the 6 degrees of freedom shown (yaw, roll, transverse, vertical, longitudinal and pitch). CM: centre of mass, the two hexagonals (blue) are local 'ground', and the circle (green) are wire clamps.

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TABLE I. Basic Tip-Tilt Mirror suspension mechanical parameters.

Parameter	value
Effective optic diameter (inc. ring)	76.2  mm  (3")
(wire separation at the optic)	
Effective entire mass (inc. ring)	00 m
Effective optic mass (inc. ring)	~90 g
Total assembly mass	$\sim 2.3 \text{ kg}$
	50.8 mm x 9.5 mm
Optic size	$(2" \times 3/8")$
	$(2 \times 3/6)$
Transmission aperture	28 mm vertical
(diameter, centred)	$\sim 35$ mm horizontal
Beam height	101.6 mm (4")
	- ( )
Pendulum length	140 mm
Suspension wire diameter	$127 \mu m$
d-pitch	1.3 mm
d-yaw	30 mm
·	
Moment of Inertia	
$I_{\mathrm{long,pitch}}$	$30.7  \mu \mathrm{N \cdot m}^2$
$ m I_{yaw}$	$37.7 \ \mu \mathrm{N} \cdot \mathrm{m}^2$
$I_{transverse,roll}$	$64.1 \ \mu N \cdot m^2$

FIG. 1. The hexagonal (blue) 'ground' points indicate where the base of the blades are attached to the mounting structure. The suspension wires are attached to the tips of the suspension blades via a clamp, as indicated by the green circles. The other end of the suspension wire, is clamped to the aluminium ring. The mechanical parameters are listed in TABLE I.

The Tip-Tilt mirror suspension is manufactured from standard ultra-high vacuum compatible materials (aluminium and stainless steel). The overall structure is  $\sim\!\!250$  mm in height. Including a 2" mirror and four actuators, the total mass of the unit is  $\sim\!\!2.3$  kg, with a footprint of 124 mm  $\times$  95 mm. FIG. 2 shows an engineering rendering of the Tip-Tilt mirror suspension.

The Tip-Tilt Mirror suspension provides isolation and control of the mode-matching and steering mirrors in the interferometric sensing chain. The suspension can also be designed to perform two other tasks. With sufficiently strong actuators (i.e., with large enough magnets), the pitch and yaw angles of the suspended mirror can be modulated at high frequencies ( $\sim 1~\rm kHz$ ) to generate alignment error signals. The Tip-Tilt can also act as a shutter (e.g. to protect sensitive photodetectors) by rapidly applying a large alignment offset. Shutter speeds as low as  $\sim 2.5~\rm ms$  have been observed in the laboratory depending on the optical configuration.

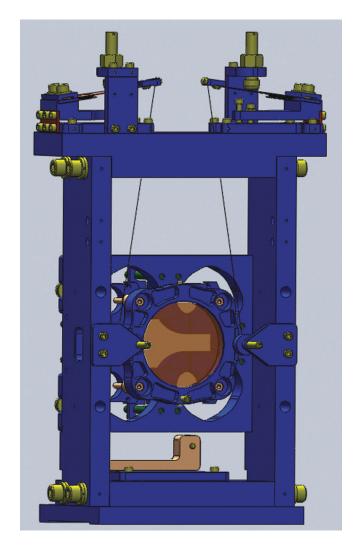


FIG. 2. An engineering rendering of the Tip-Tilt Mirror.

# A. Horizontal Isolation - Mirror Suspension

Horizontal isolation is achieved by suspending the mirror by a simple pendulum with two wires, one on either side of the aluminium ring. The wire ends are clamped between two metal plates.

The optic is clamped inside the aluminium ring by a small cylinder of Teflon (PFA-440HP) whose position is locked by a set screw. The two suspension wires are clamped onto the sides of the aluminium ring. The pendulum has a length of 140 mm and provides a longitudinal resonant frequency of 1.3 Hz. The wire is 127 µm in diameter. This diameter is thin enough to reduce flexing of the wire near the wire clamps, so that the motion associated with the wire flexing does not dominate in the optical readout, but thick enough to prevent the wire from breaking during handling and assembly.

The resonant frequency of the pitch mode, which is coupled to the longitudinal mode, is predominantly determined by the height of the wire clamping above the

TABLE II. Basic Tip-Tilt Mirror suspension modelled and measured eigenmodes  $^3$ .

Degree-of-Freedom	${\bf Modelled}$	${\bf Measured}$
Longitudinal	1.31 Hz	1.3 Hz
Pitch	$1.65~\mathrm{Hz}$	$1.63~\mathrm{Hz}$
Yaw	$1.6~\mathrm{Hz}$	$1.59~\mathrm{Hz}$
Transverse	$1.34~\mathrm{Hz}$	-
Vertical	$6.0~\mathrm{Hz}$	$5.9~\mathrm{Hz}$
Roll	$8.7~\mathrm{Hz}$	$8.6~\mathrm{Hz}$

centre of mass of the mirror assembly. This distance is denoted in FIG. 1 as d-pitch. In addition, the resonance frequency of the pitch mode is influenced by the flexing of the wire, just above the wire clamp. The wire stiffness is strongly dependent on the Young's modulus of the suspension wires, and determines the distance where the wire will flex above the wire clamp. Measurements have shown that the Young's modulus of the suspension wires can vary by almost 10%, affecting the pitch resonance by up to 1.5%.

The resonant frequency of the yaw mode is controlled by the separation of the suspension points at the top (dyaw) and the effective optic diameter. TABLE II lists the calculated fundamental eigenmodes.

## B. Vertical Isolation - Suspension Blade Design

Vertical isolation for the suspended mirror is provided by two triangular shaped beryllium copper cantilever blade springs. These blades are of a similar design to those used in the VIRGO and Advanced LIGO test mass suspensions<sup>4</sup>. In these systems, the suspension blades are made from maraging steel, pre-cured and heat treated to reduce material creep. Material creep has not been addressed in this design due to the more relaxed requirements.

Modelling of the blades was used to determine the vertical resonance and required blade parameters<sup>5,6</sup>. The blade vertical resonance frequency is given by,

$$f = \sqrt{\frac{Eah^3}{16\pi^2 m l^3 \alpha}} \quad [Hz] \tag{1}$$

where E is the Young's Modulus, a the with of the blade at the base, h the blade thickness, m the mass supported by the blade, l the length of the blade and  $\alpha$  the shape factor where a triangular blade has a shape factor of 1.5, while a rectangular blade has a shape factor of  $1^5$ .

By setting the blade thickness, the width of the base and length of the triangular suspension blade, the vertical resonance frequency was obtained. The key parameters are shown in TABLE III. The vertical resonance frequency is strongly dependent on the blade thickness and length. A practical balance was achieved between

TABLE III. The vertical blade parameters and modelling results

Parameter	set values
Base width	23 mm
Blade length	50  mm
Shape factor (triangular <sup>4</sup> )	1.5
Blade thickness	$254~\mu m$
Parameter	modelled
Vertical resonance	5.9 Hz
Effective spring constant (single blade)	$65.8 \mathrm{\ N/m}$
Deflection at tip (with 48g load)	$7~\mathrm{mm}$
Blade curvature radius (under load)	$177.3~\mathrm{mm}$
Angle of the blade-tip (under load)	$16.4 \deg$

the length of the blades (to simplify manufacturing and to keep the unit small) and the requirement to keep the vertical resonance frequency under 10 Hz.

Each blade is mounted in an aluminium structure which allows the height of the wire suspension point (clamping of the wire at the tip of the blade) to be adjusted and locked, see FIG. 3. This also provides a means of adjusting the height of the optic with respect to the coil-magnet actuators. To reduce the bending of the wire at the suspension point, the blades are launched at an angle such that the the wires leave the clamps at 90°, when they are loaded with the mirror assembly.

The vertical suspension mode has no active damping, however passive damping is achieved through the use of a 12 mm diameter viton o-ring. A small ledge is mounted under the original blade, and separated by a spacer. This ledge is a shorter version of the original blade. The viton o-ring is placed between the ledge and the suspension blade. Due to the load on the blade the o-ring is slightly compressed. Vertical motion at the blade tip, will cause the o-ring to be twisted (shear) between the ledge and

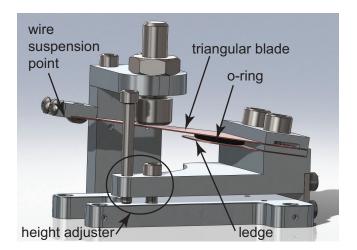


FIG. 3. Single blade units with height adjuster.

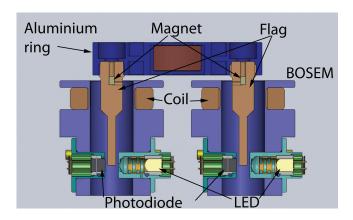


FIG. 4. A cross section of the aluminium ring with flags and the BOSEMs. Visible in the BOSEM are the photodiode, LED and coil. The flag is attached to the aluminium ring, with the magnet press fitted into it.

the blade, and provide damping that reduces the quality factor. Using this method the quality factor of the vertical mode is reduced from 180 down to 67, well below the required 100.

## III. LOCAL SENSING AND ACTUATION

Control of the mirror is achieved by four voice-coil actuators, which provide control over longitudinal, pitch and yaw degrees of freedom. Each voice-coil is paired with a magnet mounted to the suspended aluminium ring. A position sensor is also incorporated in the actuator assembly, in the form of an optical shadow sensor. A small flag is mounted to the aluminium ring. The flag is positioned such that it blocks half of the light falling on a photodiode from an opposing LED, acting as a shadow sensor, as shown in FIG. 4. Movement of the flag (and hence the aluminium ring) will be registered by the photodiode. The combination of the shadow sensor and the coil-magnet actuator provide local sensing and control of the mirror.

The shadow sensor-coil combination is built by the University of Birmingham and integrated into a single unit, referred to as the Birmingham Optical Sensor and Electro-Magnetic actuator (BOSEM)<sup>7,8</sup>. The four BOSEMs are mounted in a square with 48.22 mm sides, with the centre of the BOSEMs at a radius of 34.1 mm from the centre of the optic.

### A. Eddy Current Damping of the Eigenmodes

The longitudinal, pitch and yaw motion of the suspended mirror can be actively controlled by the four BOSEMs. However, the transverse (sideways), vertical and roll modes of the suspended mirror cannot be controlled or damped by the BOSEMs. Additional magnets

are installed to provide eddy current damping (ECD) for these modes. These smaller additional magnets ( $\varnothing 3 \text{ mm} \times 3 \text{ mm} \log$ ) are located on either side of the optic and mounted on the aluminium ring. The polarity of the magnets are arranged such that the total magnetic momentum of the mirror assembly is minimised. Eddy current damping is provided by placing small aluminium disks just in front of each of these magnets. Near-critical damping is achieved by optimising the distance between the magnets and the aluminium plates.

### B. Actuation Magnets

The standard magnets used in the BOSEMs are Nickel plated SmCo (Sm<sub>2</sub>Co<sub>17</sub>/SmCo 2:17), and are  $\varnothing 2$  mm  $\times 3$  mm long, and provide a force coefficient of 21 mN/A, in combination with the BOSEM coils. The small dimensions (e.g. magnetic strength) are chosen to mitigate the magnetic couplings with the environment. The magnets are arranged with alternating polarities, to reduce the magnetic moment of the mirror assembly.

To optimise the current-force coupling from the voice-coil to the magnet, the centre of the standard magnet needs to be  $\sim$ 6.7 mm in front of the centre of the BOSEM coil<sup>9</sup>. This is accomplished by press fitting the magnet into the back of the flags (FIG. 4).

## C. Shadow Sensor Flag

The shadow sensor flags are made from a polymer thermoplastic (PEEK) and are screwed into the mirror holder, as seen in FIG. 4. The magnet is press fitted into the flag to locate the magnet with respect to the mirror holder. The position of the BOSEM with respect to the flag may be tuned, adjusting the BOSEM along the length of the flag.

The operation point is such that the flag covers the shadow sensor at 50% of its full range with the mirror freely hanging. In this position, the distance between the BOSEM and the mirror holder is  $\sim$ 2 mm, which is sufficient to prevent it from interfering with the suspended mirror. The linear range of the BOSEM, around its operating point, is 0.7 mm<sub>pk-pk</sub><sup>8</sup>.

# IV. PERFORMANCE

The measured performance of a prototype Tip-Tilt mirror suspension was compared to numerical models as discussed below.

#### A. Transfer Functions

An electrical current in each voice coil of the BOSEMs, generates a magnetic field, which will act on the magnets

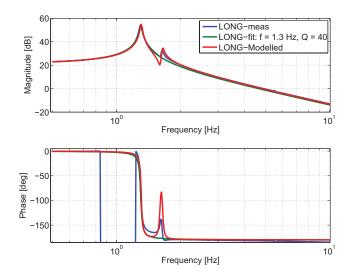


FIG. 5. Tip-Tilt longitudinal transfer function, driving the four BOSEM coils and reading out the shadow sensors.

in the aluminium ring. Applying the current through the BOSEMs in the appropriate polarity allows the aluminium ring (and optic) to be controlled in longitudinal, pitch and yaw.

FIG. 5, 6 and 7, show the longitudinal, pitch and yaw transfer functions of the Tip-Tilt mirror suspension measured in the laboratory at The Australian National University. The transfer functions were obtained by driving the coils and reading out the shadow sensors in the BOSEMs. Recorded data has been fitted with second order functions to obtain the frequency and quality factor of each resonance and plotted against the modelled response.

The fitting results are presented in the legends of the figures. The quality factors of all modes are below 100, meeting the system requirements.

# B. Pitch and Yaw Hysteresis

FIG. 8 shows two graphs of the hysteresis exhibited by the Tip-Tilt mirror suspension. The hysteresis was measured by applying a series of DC pitch (or yaw) offsets via the four BOSEM coils and recording the pitch (yaw) values sensed by the shadow sensors. After each offset step, the Tip-Tilt mirror suspension was allowed to reach equilibrium, then the sensed offset was measured with a 5 s averaging time. As shown in FIG. 8, both degrees of freedom are linear and show minimal hysteresis over a range of 8  $\rm mrad_{pk-pk}$ .

# C. Angular Pointing per Voice-Coil Current

FIG. 9 shows the pitch and yaw angle of the Tip-Tilt mirror suspension per applied BOSEM coil current.

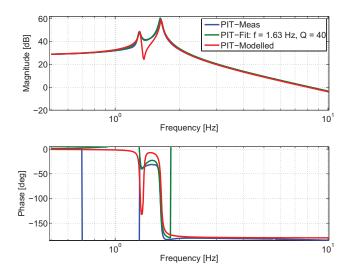


FIG. 6. Tip-Tilt pitch transfer function, driving the four BOSEM coils and reading out the shadow sensors.

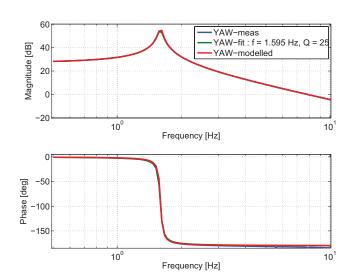


FIG. 7. Tip-Tilt yaw transfer function, driving the four BOSEM coils and reading out the shadow sensors.

This was obtained by utilising the transfer functions from FIG. 6 and 7 and applying the BOSEM optimal coilcurrent to force coefficient (modelled) to generate the graphs<sup>9</sup>. From FIG. 9, the angle-per-current coefficient below resonance is 0.05 mrad<sub>pk</sub>/mA. The maximum coil current for DC steering can be obtained by using the linear pointing range (from the hysteresis), which results in a maximum coil current of 80 mA<sub>pk</sub>. To increase the angle-to-current coefficient, larger magnets can be used.

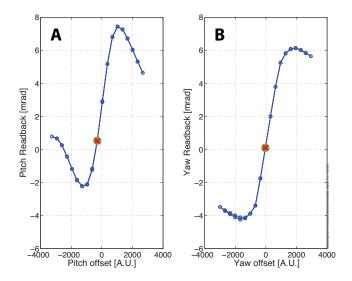


FIG. 8. Hysteresis of the Tip-Tilt mirror suspension measured using the shadow sensors in the BOSEMs, (A) Pitch Hysteresis and (B) Yaw Hysteresis. The start position is indicated by the green dot. Offsets were first increased to a positive maximum before cycling through a minimum and back to zero. The red cross indicates the end position. The turn-arounds of the measurements are due to the sensor limits, and do not represent the actual motion of the optic.

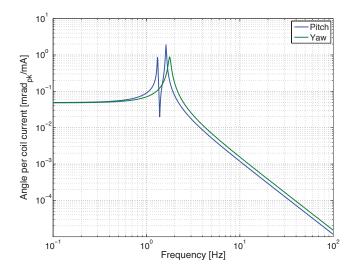


FIG. 9. Tip-Tilt mirror suspension pitch and yaw angle per applied coil current (modelled), using  $\varnothing 2$  mm  $\times$  3 mm long magnets.

## D. Vertical Mode Damping

FIG. 10 shows the performance of the o-ring damper for the vertical isolation of a single blade. The measurement was obtained by creating a shadow sensor at the wire clamp at the tip of the suspension blades using a small laser and a photodiode. The wire clamp blocked

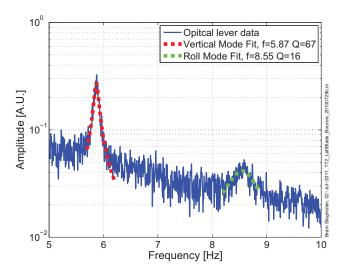


FIG. 10. Measurement of the viton o-ring damped vertical blade resonance (bounce) as well as the roll mode resonance of a single suspension blade (while installed within the Tip-Tilt mirror suspension assembly). Included in the graph is the fitted model with the results listed in the legend. The o-ring damping has reduced the vertical mode resonance to 67 (down from 180).

the laser approximately at 50% in front of the photodiode. Changes in intensity on the photodiode, due to the motion of the wire clamp, were recorded. The mirror assembly was excited by ambient air turbulence and seismic noise in the laboratory.

The figure indicates that the quality factor of the damped vertical resonance is 67, down from 180 without the damper (data not shown). In addition to damping the vertical mode, the roll mode of the suspended mirror is also damped.

# E. Eigenmodes of the Tip-Tilt Structure

The Tip-Tilt structure is designed to be stiff so that its mechanical modes will have a minimal impact on the platform to which it is mounted; the design goal was a lowest structural mode eigenfrequency of 150 Hz<sup>10,11</sup>.

The Tip-Tilt structure was mounted horizontally on a vertical actuated vibration table (TIRA TV 55240/LS-340). The Tip-Tilt structure was rotated such that the vertical motion of the structure was in the beam direction (backwards and forwards motion of the structure, see FIG. 2) and rotated by 90° for the transverse motion (sideways motion of the structure). Three accelerometers were used to record the vibrations: one was mounted on the base of the vibration platform, one on the base of the Tip-Tilt mounting bracket and one on top of the Tip-Tilt structure. The results, shown in FIG. 11, were obtained by taking the ratios between the three accelerometer measurements.

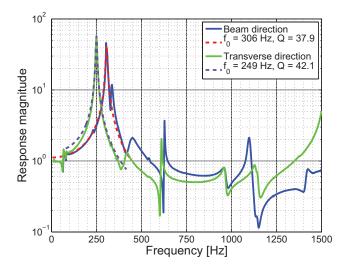


FIG. 11. Measured modal response of the Tip-Tilt structure.

The principal resonances were found to be at  $\sim 300~\mathrm{Hz}$  along the beam direction and at  $\sim 250~\mathrm{Hz}$  in the transverse direction, well above the 150 Hz target. Both resonances have a quality factor of around 40, as can be seen in FIG. 11. Due to the limitations of the available apparatus, low frequency modes (below  $\sim 5~\mathrm{Hz}$ ) were not explored.

# V. CONCLUSION

We present the design and the measured performance of the Tip-Tilt mirror suspension used in the Advanced LIGO gravitational wave detectors. These suspended steering mirrors are used in the optical chain in the interferometric sensing and control system. The six eigenmodes of the suspended mirror are all below 10 Hz, and have quality factors well below 100. The suspended mirror is controlled in longitudinal, pitch and yaw degrees of freedom via optical sensors and electro-magnetic actuators. Due to the small and sturdy design of the Tip-tilt mirror suspension, they can be useful in other optics experiments.

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